

<p>1. Approving Civil Aviation Authority/Country: FAA/United States</p>	<p>2. <b>AUTHORIZED RELEASE CERTIFICATE</b> FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG</p>				<p>3. Form Tracking Number: 22521.01</p>
<p>4. Organization Name and Address: Triumph Engine Control Systems, LLC, Charter Oak Boulevard, West Hartford, CT 06133-0651 Repair Station Number: SF3R839L</p>					
<p>6. Item:</p>	<p>7. Description:</p>	<p>8. Part Number:</p>	<p>9. Quantity:</p>	<p>10. Serial Number:</p>	<p>11. Status/Work:</p>
1.	E.T.P.U.	ETPU100-03C	1	BX36290	Repaired
<p>12. Remarks TSN: UNK CSN: UNK Temp. Rev.: 1. Work Description: Work Description:(Report bottom)(8130-Box 13-Line 1) a. Performed "Repair-6" IAW CMM. Marked &amp; etched date: (02-2004,05-2009,03-2018) b. Replaced peeling labels. c. Outgoing unit passed Vibration test. d. Unit then passed Cold, Hot, and Ambient Functional tests. e. Re-Certified unit for service. 2. Component Maintenance Manual: 75-32-19 REV 17 DTD 10/23/15 3. Service Bulletin(s): 4. Comments: Engine Serial Number: 31481</p>					
<p>TECS certifies that the work specified in block 11/12 was carried out in accordance with EASA Part 145 and in respect of that work, the [product/article] is considered ready for release to service under EASA Part 145 approval no. EASA.145.4312.</p>					
<p>13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation <input type="checkbox"/> Non-approved design data specified in Block 12</p>					
<p>14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.</p>					
<p>13b. Authorized Signature:</p>			<p>14b. Authorized Signature</p>		
<p>13c. Approval Authorization No.:</p>			<p>14c. Approval/Certificate No: SF3R839L</p>		
<p>13d. Name (Typed or Printed):</p>			<p>14d. Name (Typed or Printed): RON RINEHART</p>		
<p>13e. Date (dd/mm/yyyy):</p>			<p>14e. Date (dd/mm/yyyy): 19 Apr 2018</p>		
<p><b>User/Installer Responsibilities</b></p>					
<p>It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer performs work in accordance with the national regulation. For work which is performed by an authority different from the airworthiness authority of the country specified in block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s) article(s) from the airworthiness authority of the country specified in Block 1. Statements in block 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certificate issued in accordance with the national regulations by the user/installer before the aircraft may be flown.</p>					

### MAINTENANCE REPORT



Triumph Engine Control Systems

A Honeywell Group Company

Repair Station Air Agency  
Certificate Number

SF3R839L

Customer Name:  
TEMCO LIMITED

Report Number: 22521.01

Operator Name:  
TEMCO

Date Initiated: 03/17/2018

TECS P/N: ETPU100-03C  
TECS P/N Output: 100-03C

Component/Accessory Nomenclature:

Serial Number:

E.T.P.U.

BX36290

Customer P/N:

Customer P/O No: R2778

TSN:UNK

CSN:UNK

CSO:

CMM: 75-32-19 REV 17 DTD 10/23/15

Date Received: 3/15/2018

Removal Date:

TECS S/O No: 00537310

Work Order: 00851458

Service Bulletins Incorporated This Visit:

Customer Instructions:

Confirmed? N/A

#### Reason for Return:

TRANSIENT PRESSURE

#### Component / Accessory Disposition:

- A. Fault Codes: CC,AA(4),FA,40,ED / Flags: All within normal range. Warning label, BITE label and switch label peeling off.
- B. Unit passed: Insulation & Zener tests.  
Unit passed: Ambient, Cold & Hot functional tests.
- C. All "plug in" modules checked for defects, none found.
- D. History checked. "Repair-6" required, last done: 05-2009.

#### Summary of Work:

- Work Description:(Report bottom)(8130-Box13-Line 1)
- a. Performed "Repair-6" IAW CMM. Marked & etched dates:(01-2004,05-2009,05-2014)
  - b. Replaced peeling labels.
  - c. Outgoing unit passed Vibration test.
  - d. Unit then passed Cold, Hot, and Ambient Functional tests.
  - e. Re-Certified unit for service.

#### Comments / Warranty Statement:

Written By: RON RINEHART

Date Completed: 4/19/2018

Nothing in this report shall be deemed to be an admission by the Company of liability for any loss or damage of any kind whatsoever caused directly or indirectly by a defect in material or by any defect in equipment the subject of this report or by any negligence of the Company or any servant or agent of the Company Triumph Engine Control Systems West Hartford CT USA.

THIS REPORT CONTAINS NO EXPORT CLASSIFIABLE TECHNICAL DATA.

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