Stax ment regr ations	Whe the	It is impor			13d. Nam		13 b. Autr		13a, Cr	comp	Soniq	Unit h	12. Remarks:		6. Item:	4. Org	FA	 ⊊_
Star ments in Blocks 13a and 14a do not constitute installation or reproations by the user/installer before the aircraft may be flown.	us. vinstaler performs wo	nt to nderstand that the			13d. Name (Typed or Printed)		13 b. Authorized Signature:	Approved design date Non-approved design	Certifies the items identified above were manufactured in conformity to	onent is considered r	Soniq Aerospace certifies that the work specified in Block 11/12 was carried out in accordance with Excomponent is considered ready for release to service under EASA Part-145 Approval pum, er LASA 1	Unit has been Overhauled 01JUL2019. See Teardown listed in hox 5	marks:	Window AY -	7. Description	4. Organization Name and Address:	FAA / United States	<ol> <li>Approving Civil Aviation Authority/Country</li> </ol>
o not constitute installatione the aircraft may be flow	ork in accordance with the is/her airworthiness autho	existence of this docume						Approved design data and are in a condition frresafe operation of the condition frresafe operation in Block 12.	l above were manufacture	eady for release to s	that the work specific	, Inspected and Teston Report for the perti		Window AY - Openable LH				, i
Star ments in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an instance with the contain an instance of the contain and instance of the contain an instance of the contain and instance of	national regulations of an airworth	nt alone does not automatically co	User/Inst	13e: Date (dd/mmmi/yyyyy);		13c Aproval/Authorization No.	200 (2000) 2000   10 miles   10 m	d in conformity to:	component is considered ready for release to service under EASA Part-145 Approv⊲ num. er L	ed in Block 11/12 was carrie	ed in accordance with the la		141A4810-53	8. Part Number	Soniq Aerospace, LP 175 Roy Road SW, Bldg A, Suite 108 Pacific, WA 98047	FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	AUTHORIZED RE	
maintenance reco	iness authority diffu	stitute authority to	User/Installer Responsibilities	Dusti	14d. Na		14b. Auth	2 7 6 0	14a.	Approval num	d out in accorda	test Manufactu er. Full details h		<b></b> 3.	9. Quantity:	FAA Repair Station EASA Reference: Ph: 253-200-4991	ORTHINESS,	I FASE C
ords must contain an installation certificati	Whe \the us. \(installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/i \(\text{aller ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.	is import not to aderstand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.	litles	14d. Name (Typed or Printed)  Dustin Ronk	me (Typed or Printed)	14b. Authorized Signature:	Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title Block 11 and the scordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.	14 CFR 43.9 Return to Service	er L 1SA 1 ,5.6389.	ance with E. SA, art-145 and in	Unit has been Overhauled, Inspected and Tested in accordance with the latest Manufacturer's data, Boring \MM 56-10-02, Revision No. 41, Dated 01JUL2019. See Teardown Report for the pertinent details of this work order. Full details held on file at he report station under the Work order Number listed in box 5.		WN026	10 Serial Number:	FAA Repair Station: 90AR440B EASA Reference: EASA 145.6389 Ph: 253-200-4991	RELEASE CERTIFICATE AIRWORTHINESS APPROVAL TAG		
on issued in accor	he country specifi itry specified in Bl	9.	And the second s		14e. [	`	14c. A	n Block 12, the wo accomplished in a and in respect to	Other regu	opposit to tist	respect to that	2, Revision No			4	5. Wo		3. For
tallation certification issued in accordance with the national	ied in Block 1, it is essential lock 1.		The state of the s	16/Oct/2019	Date (dd/mmm/yyyy)	90AR440B	Approval/Certificate No:	erwise specified in Block 12, the work identified in in Block 12 was accomplished in accordance with Title julations, part 43 and in respect to that work, the items to service.	Other regulation specified in Block 12		twork the	5, 41, Dated corder Number		Overhauled	State & Work:	Work Order/Cc tract/Invoice	19227-1	3. Form Tracking Number



Teardown Report

Soniq Aerospace, LP
FAA Repair Station # 90AR440B
175 Roy Rd. SQ, Bldg A, Suite 108
Pacific, WA 98047
T: 253-200-4991
F: 253-750-4592
www.sonigaero.com www.soniqaero.com

WO#:	19227-1
Date:	160072019

Customer:	onic * erospace	Cust Phone #:	
Description:	Wir dow ^:: - Openable LH	PO#	For Stock
Part Number:	141A481 ~53	Due Date:	10/16/2019
Serial Number	WN0263	Aircraft Make/Mo	del B737NG
Receiving Inspe Unit r 7176	the second secon	I serial numbers for accuracy. Customer traceat 1A4810-14 SN WN0263.	bility documentation states PN 5-
Perfo	pair Instructions: rm minor modification from 1 ،1A4810 / ical data and release to servi. > with _ual	to 141A4810-53 and overhaul in accordance w release 8130-3.	ith the manufacturer's current
Tech Data Used	Boeing CMM 56-10-02, Rev 41, Da	ited 01JUL2019	
14 to	panel, 5-89355-87 has vinyl fogging due ing. Pressure seal tom. Lower forward fra 141A4810-53.	to, ic sture ingression and is beyond repair. Fr ame lece of maged. Unit requires overhaul an	ame assembly worn. Bearings require d minor modification from 141A4810-
	Found		
Functional Tes For Ro Senso	ts : eplacement Panel: 5-89355-87 SN 14169 ors: 306 OHMS, 306 OHMS		o Bus: 59.4 OHMS mp: 68 Degrees Fahrenheit
Discrepancies: Per Pr	reliminary Inspection.		
Corrective Action			
Disass modifi compo	sembled unit. Stripped and inspected frai cation from 141A4810-14 to 141A4810-5 pnents.Installed glass panel 5-89355-87	me and mechanism components for do lage, not also also also also also also also also	ame and mechanism
Parts Used:			
5-8935 Primei	55-87 SN 14169H9333, Panel AY; 9-6652 r and Top Coat.	5-13, Pressure Seal; 141A4811-5, Frame; Mis-	r ,1ardware र าd Bearings; Sealant,
Scrapped Parts:			
Pressu	ure Seal; 141A4811-5, Frame; Misc Hard	ware and Bearings	The state of the s
inal Inspection:			THE A CHIEF PROPERTY OF THE STATE OF THE STA
Unit wa approv	as overhauled, functional tested and insp red for return to service.	ected in accordance with Boeing CMM 56-10-0	2, Rev 41, Date or JUL 119 and is
	$\supset$		
<u>·</u>	16007201	· \$	
Inspector's Sig	nature Date	•	